

Reynold's transport theorem

Let a velocity field $\vec{v}(t, \vec{x})$ be given in a subset $D \subset \mathbb{R}^3$. Let $R \subset D$, and for each $\vec{x}_0 \in R$, let $\vec{x}(t, \vec{x}_0)$ be the solution to

$$\frac{d\vec{x}}{dt} = \vec{v}(t, \vec{x}), \quad \vec{x}(0) = \vec{x}_0$$

thus the curve $\vec{x}(t, \vec{x}_0)$ "flows along" the velocity field. If we let $R = R(0)$, we define

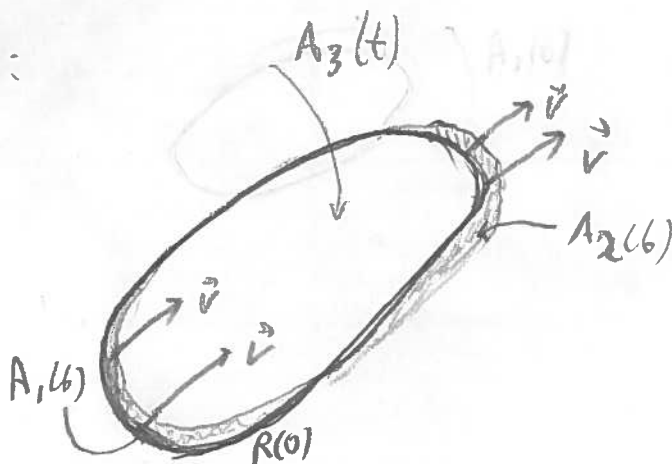
$$R(t) = \{ \vec{x}(t; \vec{x}_0) : \vec{x}_0 \in R(0) \}.$$

Reynolds transport theorem then states

$$\frac{d}{dt} \iiint_{R(t)} f(t, \vec{x}) dV = \frac{d}{dt} \iiint_{R(0)} f(t, \vec{x}) dV + \iint_{\partial R(0)} f(t, \vec{x}) \vec{v} \cdot \vec{n} dS$$

where \vec{n} points out of $R(0)$.

Proof:



Define:

$$A_1(t) \equiv R(t) \setminus R(0)$$

$$A_2(t) = R(0) \setminus R(t)$$

$$A_3(t) = R(0) \cap R(t)$$

Then

$$R(0) = A_1(t) \cup A_3(t)$$

$$R(t) = A_2(t) \cup A_3(t)$$

$A_1(t), A_2(t), A_3(t)$ are mutually disjoint.

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Also define: $S_1 = \{ \vec{x} \in \partial R(t) : \vec{v}(t, \vec{x}) \cdot \vec{n} < 0 \}$

$S_2 = \{ \vec{x} \in \partial R(t) : \vec{v}(t, \vec{x}) \cdot \vec{n} \geq 0 \}$

So, $\partial R(t) = S_1 \cup S_2$.

Finally define $\mathcal{O}_j(t) = \iiint_{A_j(t)} f(t, \vec{x}) dV$, $j=1, 2, 3$.

So $\iiint_{R(t)} f(t, \vec{x}) dV = \mathcal{O}_2(t) + \mathcal{O}_3(t)$, and

$\iiint_{R(t)} f(t, \vec{x}) dV = \mathcal{O}_1(t) + \mathcal{O}_3(t)$

We then have

$$\frac{d}{dt} \iiint_{R(t)} f(t, \vec{x}) dV = \lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} \left(\iiint_{R(t+\Delta t)} f(t+\Delta t, \vec{x}) dV - \iiint_{R(t)} f(t, \vec{x}) dV \right)$$

$$= \lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} \left(\mathcal{O}_2(t+\Delta t) + \mathcal{O}_3(t+\Delta t) - \mathcal{O}_3(t) \right)$$

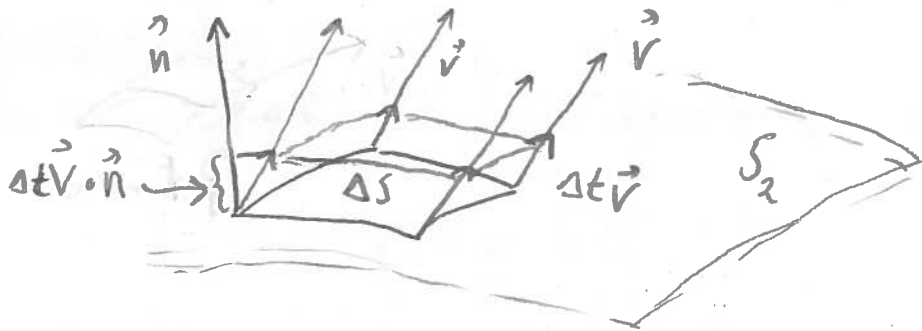
$$= \lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} \left[\left(\mathcal{O}_3(t+\Delta t) + \mathcal{O}_1(t+\Delta t) - \mathcal{O}_3(t) \right) + \left(\mathcal{O}_2(t+\Delta t) - \mathcal{O}_1(t+\Delta t) \right) \right]$$

$$= \lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} \left[\iiint_{R(t+\Delta t)} f(t+\Delta t, \vec{x}) dV - \iiint_{R(t)} f(t, \vec{x}) dV \right]$$

$$+ \lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} \left[\mathcal{O}_2(t+\Delta t) - \mathcal{O}_1(t+\Delta t) \right] =$$

$$= \frac{d}{dt} \iiint_{R(0)} f(t, \vec{x}) dV \Big|_{t=0} + \lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} (\mathcal{Q}_2(\Delta t) - \mathcal{Q}_1(\Delta t)) \quad (3)$$

On S_2 we have $\vec{v} \cdot \vec{n} \geq 0$. During time $[0, \Delta t]$ a point on S_2 moves $\Delta t \vec{v}$ out from $R(0)$.



View $A_2(\Delta t)$ as a thin layer over S_2 with thickness $\Delta t \vec{v} \cdot \vec{n}$, where \vec{n} is unit normal vector to S_2 .

$$\begin{aligned} \text{Thus, } \mathcal{Q}_2(\Delta t) &= \iiint_{A_2(\Delta t)} f(\Delta t, \vec{x}) dV \approx \iint_{S_2} f(\Delta t, \vec{x}) (\Delta t \vec{v}) \cdot \vec{n} dS \\ &= \Delta t \iint_{S_2} f(\Delta t, \vec{x}) \vec{v} \cdot \vec{n} dS \Rightarrow \end{aligned}$$

$$\lim_{\Delta t \rightarrow 0} \frac{\mathcal{Q}_2(\Delta t)}{\Delta t} = \iint_{S_2} f(0, \vec{x}) \vec{v} \cdot \vec{n} dS.$$

On S_1 , $\vec{v} \cdot \vec{n} < 0$ and $A_1(\Delta t)$ is a thin layer over S_1 with thickness $-\Delta t \vec{v} \cdot \vec{n}$. And as with $\mathcal{Q}_2(\Delta t)$

we have

$$\mathcal{Q}_1(\Delta t) \approx -\Delta t \iint_{S_1} f(\Delta t, \vec{x}) \vec{v} \cdot \vec{n} dS.$$

We have shown that

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$$\lim_{\Delta t \rightarrow 0} \frac{1}{\Delta t} (\mathcal{O}_2(\Delta t) - \mathcal{O}_1(\Delta t)) = \iint_{S_2} f(\mathbf{0}, \vec{x}) \vec{v} \cdot \vec{n} dS + \iint_{S_1} f(\mathbf{0}, \vec{x}) \vec{v} \cdot \vec{n} dS$$
$$= \iint_{dR(\mathbf{0})} f(\mathbf{0}, \vec{x}) \vec{v} \cdot \vec{n} dS,$$

and the proof is complete.

Navier-Stokes equations, fluid mechanics

We consider flow of a fluid (liquid or gas) in 3 spatial dimensions. At each time and position (t, \vec{x}) the fluid is assumed to have a well defined mass density: $\rho = \rho(t, \vec{x})$

velocity = $\vec{v} = \vec{v}(t, \vec{x}) = [v_1(t, \vec{x}), v_2(t, \vec{x}), v_3(t, \vec{x})]$

energy = $e = e(t, \vec{x})$

We will obtain 5 equations for these 5 unknowns. The 5 equations express

- 1) mass conservation
- 2) energy conservation
- 3) 3 equations for momentum conservation.

Mass conservation

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Let $R(t)$ be a domain following the flow. Then, since no mass is created or lost, we have

$$\frac{d}{dt} \iiint_{R(t)} \rho \, dV = 0$$

Using Reynolds transport theorem we obtain

$$\frac{d}{dt} \iiint_{R(t)} \rho \, dV + \iint_{\partial R(t)} \rho \vec{v} \cdot \vec{n} \, dS = 0 \quad (1)$$

Moving $\frac{d}{dt}$ inside the integral, and using the divergence theorem, we have

$$\iiint_{R(t)} \left(\frac{\partial \rho}{\partial t} + \vec{v} \cdot (\rho \vec{v}) \right) dV = 0$$

Since $R(t)$ is arbitrary (and $\frac{\partial \rho}{\partial t} + \vec{v} \cdot (\rho \vec{v})$ continuous) we have

$$\frac{\partial \rho}{\partial t} + \vec{v} \cdot (\rho \vec{v}) = 0 \quad (2)$$

(2) is mass conservation on differential form, while (1) is this law on integral form.

Note that (2) is non-linear as it contains the product $\rho \vec{v}$ between unknowns.

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Special case: Incompressible flow.

When the fluid is incompressible, the mass density is constant. So, $\frac{d\rho}{dt} = 0$ giving

$$\vec{\nabla} \cdot (\rho \vec{v}) = 0 \Leftrightarrow \rho \vec{\nabla} \cdot \vec{v} = 0 \Leftrightarrow \underline{\vec{\nabla} \cdot \vec{v} = 0}$$

Conclusion: For incompressible fluids we must have

$$\boxed{\vec{\nabla} \cdot \vec{v} = 0}$$

This is very reasonable as $\vec{\nabla} \cdot \vec{v} = 0 \Rightarrow 0 = \iiint_R \vec{\nabla} \cdot \vec{v} \, dV$

Divergence theorem $\Leftrightarrow 0 = \iint_{\partial R} \vec{v} \cdot \vec{n} \, dS$, i.e. the net flow

out of R is zero. If it was not, mass would accumulate (or the opposite) and the density change.

Momentum conservation

Momentum conservation is really Newton's 2-law

"force = mass · acceleration", that is,

$$\vec{F} = \frac{d}{dt} m \vec{v} \quad \text{for a rigid body with mass } m,$$

where $m \vec{v}$ is the (linear) momentum of the body.

We define the momentum density as

$$\rho \vec{v},$$

and the momentum of a region $R(t)$ going with the flow as

$$\iiint_{R(t)} \rho \vec{v} dV$$

Conservation of momentum is

$$\frac{d}{dt} \iiint_{R(t)} \rho \vec{v} dV = \sum \vec{F} = \text{sum of all forces acting on } R(t)$$

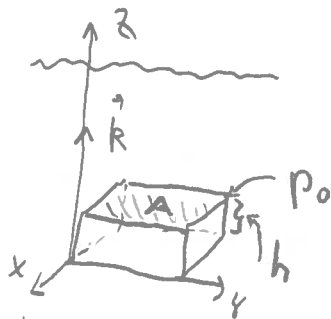
source for momentum

$$\text{or } \frac{d}{dt} \iiint_{R(t)} \rho \vec{v} dV = \vec{F}^B + \vec{F}^S$$

where \vec{F}^B is body force, and \vec{F}^S surface force.

\vec{F}^B is typically gravity forces, while \vec{F}^S is the force the surrounding fluid acts with on the surface of $R(t)$.

Example



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A box with density ρ_1 and volume Ah is submerged in a liquid with density ρ_2 . What is the total force acting on the box?

Body force is gravity; $\vec{F}^B = -\rho_1 h A g \vec{k}$, g acceleration of gravity

Surface force from pressure:

Pressure acts on the surface of the box, with a force in the opposite direction of the outpointing unit normal \vec{n} .

By common reason the forces on the 4 vertical sides of the box cancel out.

On the top horizontal side, the force is

$$-p_0 A \vec{k}, \text{ (downward)}$$

where p_0 is the pressure at this depth.

On the bottom side the hydrostatic pressure is

$$p_0 + \rho_2 g h, \text{ giving force}$$

$$(p_0 + \rho_2 g h) A \vec{k} \text{ acting upwards.}$$

Thus

$$\begin{aligned}\vec{F}^S &= A(\rho_0 + \rho_2 gh) \vec{k} - \rho_0 A \vec{k} \\ &= \underline{(\rho_2 h A) g \vec{k}} \quad (\text{Archimedes' law})\end{aligned}$$

Thus, the total force is

$$\vec{F} = \vec{F}^B + \vec{F}^S = (\rho_2 - \rho_1) gh.$$

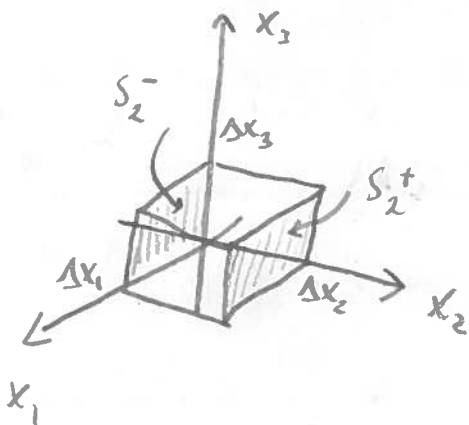
Surface forces

The so-called stress tensor $\sigma = \sigma(t, \vec{x})$

is a 3×3 -matrix, defined such that

$$(\sigma \vec{n}) \Delta S$$

is the surface force acting on a small surface ΔS with unit normal \vec{n} .



Thus, the force in the i -direction on surface S_2^+

$$\text{is } \sigma_{i2}(x_1, \Delta x_2, x_3) \Delta x_1 \Delta x_3$$

Since \vec{n} has opposite direction on S_2^- , the force in the i -direction

$$\text{on } S_2^- \text{ is } -\sigma_{i2}(x_1, 0, x_3) \Delta x_1 \Delta x_3.$$

So the net force in the \hat{i} -direction on S_2^+ and S_2^- is.

$$\begin{aligned}
& (\sigma_{i2}(x_1, \Delta x_1, x_3) - \sigma_{i2}(x_1, 0, x_3)) \Delta x_1 \Delta x_3 \\
& \approx \left(\frac{d\sigma_{i2}}{dx_2}(x_1, x_2^0, x_3) \Delta x_2 \right) \Delta x_1 \Delta x_3 \quad \text{for } x_2^0 \in [0, \Delta x_2] \\
& = \frac{d\sigma_{i2}}{dx_2}(x_1, x_2^0, x_3) \Delta V.
\end{aligned}$$

Consequently, the total surface force in the \hat{i} -direction is

$$\left(\sum_{j=1}^3 \frac{d\sigma_{ij}}{dx_j} \right) \Delta V,$$

Thus, the surface force in the \hat{i} -direction on $R(\delta)$ is

$$\iiint_{R(\delta)} \left(\sum_{j=1}^3 \frac{d\sigma_{ij}}{dx_j} \right) dV$$

By defining $\vec{\nabla} \cdot \sigma$ as the vector with component i :

$$(\vec{\nabla} \cdot \sigma)_i = \sum_{j=1}^3 \frac{d\sigma_{ij}}{dx_j},$$

we find
$$\vec{F}^s = \iiint_{R(\delta)} \vec{\nabla} \cdot \sigma dV$$

Defining \vec{f}^B as

$$\vec{F}^B = \iiint_{R(b)} \vec{f}^B dV,$$

i.e. \vec{f}^B is the body force density + we obtain

$$\frac{d}{dt} \iiint_{R(b)} \rho \vec{v} dV = \iiint_{R(b)} (\vec{f}^B + \vec{\nabla} \cdot \underline{\sigma}) dV \quad (6)$$

Cauchy stress tensor

$$\text{deb } \sigma_{ij} = -\delta_{ij} p + \tilde{\sigma}_{ij},$$

where $\delta_{ij} = \begin{cases} 1 & \text{if } i=j \\ 0 & \text{if } i \neq j \end{cases}$ is the Kronecker delta symbol.

The $-\delta_{ij} p$ part, p pressure, acts always normal to the surface, while the $\tilde{\sigma}_{ij}$ part represent the friction forces.

How do we define $\tilde{\sigma}_{ij}$? If $\tilde{\sigma}$ is to represent frictional effects, it should depend on the variation in space of the field variables ρ, \vec{v} , and e .

, i.e. depend on the various derivatives

$$\frac{\partial p}{\partial x_j}, \frac{\partial e}{\partial x_j}, \frac{\partial v_i}{\partial x_j} \quad i, j = 1, 2, 3.$$

Now, this is a fact: In order for our equations to be valid in all inertial systems (Galilean invariance)

$\tilde{\sigma}$ cannot depend on $\frac{\partial p}{\partial x_j}, \frac{\partial e}{\partial x_j}$.

To our level of approximation, we assume $\tilde{\sigma}$

is linear in $\frac{\partial v_i}{\partial x_j}, i, j = 1, \dots, 3$, i.e.

$$\tilde{\sigma}_{ij} = \sum_{mn} \alpha_{ijmn} \frac{\partial v_m}{\partial x_n} \quad \alpha_{ijmn} \text{ constants.}$$

Further physical restrictions (we have an isotropic fluid) narrow $\tilde{\sigma}$ down to

$$\tilde{\sigma}_{ij} = a \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right) + b \delta_{ij} \vec{\nabla} \cdot \vec{v},$$

for some constants a, b .

We want to split the matrix $\tilde{\sigma}$ in two terms such that the first term has trace zero.

We see that

$$\begin{aligned} & \text{trace} \left\{ a \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right) \right\}_{i,j} \\ &= \sum_{i=1}^3 a \left(\frac{\partial v_i}{\partial x_i} + \frac{\partial v_i}{\partial x_i} \right) = 2a \vec{v} \cdot \vec{v} \\ & \text{trace} \{ b \delta_{ij} \vec{v} \cdot \vec{v} \} = 3b \vec{v} \cdot \vec{v} \end{aligned}$$

Thus writing,

$$\tilde{\sigma}_{ij} = \underbrace{a \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} - \frac{2}{3} \delta_{ij} \vec{v} \cdot \vec{v} \right)}_{\text{traceless}} + \left(b + \frac{2}{3}a \right) \delta_{ij} \vec{v} \cdot \vec{v}$$

we obtain the final expression for $\tilde{\sigma}_{ij}$:

$$\tilde{\sigma}_{ij} = \eta \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} - \frac{2}{3} \delta_{ij} \vec{v} \cdot \vec{v} \right) + \zeta \delta_{ij} \vec{v} \cdot \vec{v}$$

Hence $\eta = a$, $\zeta = b + \frac{2}{3}a$ are the shear viscosity and the bulk (or volume) viscosity, respectively.

The shear viscosity η is the "standard" viscosity of the fluid. While the bulk viscosity ζ is often disregarded, but can be important for rapidly expanding gases, and ζ plays a role in damping of sound waves.

Recall the conservation equation for momentum (6):

$$\frac{d}{dt} \int_{RV} \rho \vec{v} dV = \int_{RV} \vec{f}^B dV + \int_{RV} \vec{v} \cdot \sigma dV$$

Thus, we need to calculate $\vec{\nabla} \cdot \sigma$:

We have $(\vec{\nabla} \cdot \sigma)_i = \sum_{j=1}^3 \frac{d\sigma_{ij}}{dx_j}$ and

$$\sigma_{ij} = -\delta_{ij}p + \eta \left(\frac{dU_i}{dx_j} + \frac{dU_j}{dx_i} - \frac{2}{3} \delta_{ij} \vec{\nabla} \cdot \vec{v} \right) + \zeta \delta_{ij} \vec{\nabla} \cdot \vec{v}$$

Thus, $(\vec{\nabla} \cdot \sigma)_i = -\frac{dp}{dx_i} + \eta \left[\sum_{j=1}^3 \left(\frac{d^2 U_i}{dx_j^2} + \frac{d^2 U_j}{dx_i dx_j} \right) - \frac{2}{3} \frac{d}{dx_i} (\vec{\nabla} \cdot \vec{v}) \right]$

$$+ \zeta \frac{d}{dx_i} (\vec{\nabla} \cdot \vec{v})$$

$$= -\frac{dp}{dx_i} + \eta \left[\nabla^2 U_i + \frac{d}{dx_i} \left(\vec{\nabla} \cdot \vec{v} - \frac{2}{3} \vec{\nabla} \cdot \vec{v} \right) \right]$$

$$+ \zeta \frac{d}{dx_i} (\vec{\nabla} \cdot \vec{v})$$

So, we see

$$\vec{\nabla} \cdot \sigma = -\vec{\nabla} p + \eta \left[\nabla^2 \vec{v} + \frac{1}{3} \vec{\nabla} (\vec{\nabla} \cdot \vec{v}) \right] + \zeta \vec{\nabla} (\vec{\nabla} \cdot \vec{v})$$

Thus, the right hand side of (6) is:

$$\iiint_{R(t)} \left(\vec{f}^b - \vec{\nabla} p + \eta \left[\nabla^2 \vec{v} + \frac{1}{3} \vec{\nabla} (\vec{\nabla} \cdot \vec{v}) \right] + \zeta \vec{\nabla} (\vec{\nabla} \cdot \vec{v}) \right) dV \quad (7)$$

For the left hand side of (6) we use Reynolds transport theorem:

theorem: $\frac{d}{dt} \iiint_{R(t)} \rho v_i dV = \iiint_{R(t)} \frac{d}{dt} (\rho v_i) dV + \iint_{\partial R(t)} \rho v_i \vec{v} \cdot \vec{n} dS$
 $i=1,2,3$

$$= \iiint_{R(t)} \left(\frac{d}{dt} (\rho v_i) + \vec{\nabla} \cdot (\rho \vec{v}) \right) dV$$

$$= \iiint_{R(t)} \left(\frac{d\rho}{dt} v_i + \rho \frac{dv_i}{dt} + \rho_i \vec{\nabla} \cdot (\rho \vec{v}) + \rho \vec{v} \cdot (\vec{\nabla} v_i) \right) dV$$

Since $\vec{\nabla} \cdot (\rho \vec{v}) = - \frac{d\rho}{dt}$ from mass conservation, we obtain

$$\iiint_{R(t)} \rho \left(\frac{dv_i}{dt} + \vec{v} \cdot (\vec{\nabla} v_i) \right) dV, \text{ or on vector form}$$

$$\iiint_{R(t)} \rho \left(\frac{d\vec{v}}{dt} + (\vec{v} \cdot \vec{\nabla}) \vec{v} \right) dV. \text{ Thus, from (7) we}$$

arrive at the final equation:

$$\rho \left(\frac{d\vec{v}}{dt} + (\vec{v} \cdot \vec{\nabla}) \vec{v} \right) = -\vec{\nabla} p + \eta (\nabla^2 \vec{v} + \frac{1}{3} \vec{\nabla} (\vec{\nabla} \cdot \vec{v})) + \zeta \vec{\nabla} (\vec{\nabla} \cdot \vec{v}) + \vec{f}^B$$

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Special cases: Incompressible fluid, $\rho = \text{const.}$

$$\rho \left(\frac{d\vec{v}}{dt} + (\vec{v} \cdot \vec{\nabla}) \vec{v} \right) = -\vec{\nabla} p + \eta \nabla^2 \vec{v} + \vec{f}^B$$

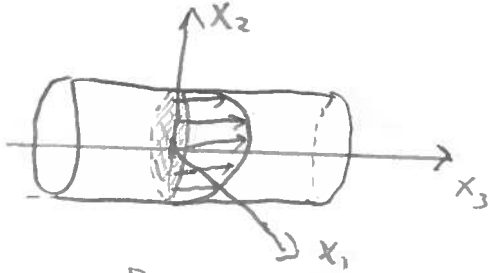
$$\Leftrightarrow \frac{d\vec{v}}{dt} + (\vec{v} \cdot \vec{\nabla}) \vec{v} = -\frac{1}{\rho} \vec{\nabla} p + \left(\frac{\eta}{\rho}\right) \nabla^2 \vec{v} + \frac{1}{\rho} \vec{f}^B$$

Inviscid fluid $\eta = \zeta = 0$

$$\rho \left(\frac{d\vec{v}}{dt} + (\vec{v} \cdot \vec{\nabla}) \vec{v} \right) = -\vec{\nabla} p + \vec{f}^B$$

Example: Laminar flow in a circular tube,

Assume steady state laminar flow (low Reynolds number) in a circular tube of radius R , aligned in direction x_3 .



All flow is in the x_3 -direction, so

$$\vec{v} = \begin{bmatrix} 0 \\ 0 \\ v_3(r) \end{bmatrix}, \text{ where } r \text{ is the polar coordinate}$$

for radius in the x_1, x_2 -plane. Since $\frac{dv_3}{dt} = 0$

(steady state), and $\vec{v} \cdot \nabla = v_3 \frac{d}{dx_3}$, we have

$$(\vec{v} \cdot \nabla) \vec{v} = v_3 \frac{d}{dx_3} \begin{bmatrix} 0 \\ 0 \\ v_3(r) \end{bmatrix} = \vec{0}$$

Assume incompressibility, and from (9) we have

$$\vec{0} = -\vec{\nabla} p + \gamma \nabla^2 \vec{v}$$

Since we have steady state flow in x_3 direction, we have

$$\vec{\nabla} p = \begin{bmatrix} 0 \\ 0 \\ p_{x_3} \end{bmatrix}, \text{ where } p_{x_3} = \frac{dp}{dx_3} = \text{const.}$$

$$\gamma \nabla^2 \vec{v} = \gamma \nabla^2 \begin{bmatrix} 0 \\ 0 \\ v_3(r) \end{bmatrix} = \gamma \begin{bmatrix} 0 \\ 0 \\ \nabla^2 v_3(r) \end{bmatrix}$$

$$\gamma \begin{bmatrix} 0 \\ 0 \\ \frac{d^2 v_3}{dx_1^2} + \frac{d^2 v_3}{dx_2^2} \end{bmatrix} = \gamma \begin{bmatrix} 0 \\ 0 \\ v_3''(r) + \frac{1}{r} v_3'(r) \end{bmatrix} \quad \begin{matrix} \nabla^2 f(r) = \\ f'' + \frac{1}{r} f' \\ \text{for polar coordinates} \end{matrix}$$

This gives

$$\begin{aligned} \rho x_3 &= \eta (v_3'' + \frac{1}{r} v_3') \\ &= \eta (\frac{1}{r} (r v_3'))' \end{aligned}$$

Thus $r v_3' = \frac{1}{2} \eta r^2 \rho x_3 + C$.

We see that $C = 0$ as $r \rightarrow 0$. This gives

$$v_3' = \frac{1}{2} r \rho x_3 / \eta$$

$$\Rightarrow v_3 = \frac{1}{4} \frac{r^2}{\eta} \rho x_3 + C_1$$

Now, the so-called no-slip condition says that the velocity at the wall of the tube is zero. This gives

$$v_3(R) = 0 \Leftrightarrow C_1 = -\frac{1}{4} \frac{R^2}{\eta} \rho x_3$$

Thus, the velocity is

$$\underline{v_3(r) = \frac{\rho x_3}{4\eta} (R^2 - r^2)}$$

We see the velocity is inversely proportional to the shear viscosity, which makes sense.

The volumetric flux through the tube is (D_R is cross-section of tube)

$$Q = \iint_{D_R} v_3(r) dA = \int_0^{2\pi} \int_0^R \left(\frac{\rho x_3}{4\eta} (R^2 - r^2) \right) r dr d\theta$$

$$= 2\pi \frac{\rho_x}{4\eta} \int_0^R \left(\frac{R^2 r^2}{2} - \frac{1}{4} r^4 \right) dr$$

$$= \frac{\pi}{8} \frac{\rho_x}{\eta} R^4$$

Thus, we can measure η

by applying a pressure drop over the tube, and measure the total flow rate Q , and find

$$\eta = \frac{\pi}{8} \frac{\rho_x}{Q} R^4$$

The volumetric (or total) flow-rate also gives us the average velocity

$$\bar{v} = \frac{Q}{A} = \frac{Q}{\pi R^2} = \frac{R^2}{8\eta} \rho_x$$

Comparing to Darcy's law

$$\bar{v} = \frac{k}{\eta} \rho_x \quad (\text{ignoring sign})$$

we find the permeability of a circular tube is

$$k = R^2/8 \quad (\text{Poiseuille's law})$$

Thus, a tube with $R = 10^{-5} \text{ m}$ has a permeability of

$$k = \frac{10^{-10}}{8} \text{ m}^2 \approx 1.25 \cdot 10^{-11} \text{ m}^2 \approx 12.5 \cdot 10^{-12} \text{ m}^2 \approx 12.5 \text{ Darcy}$$

which is a high permeability compared to the permeability of a sand stone.

Conservation of energy

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We have "energy = force · distance", or in vector form

$$dE = \vec{F} \cdot d\vec{r} = \vec{F} \cdot \vec{v} dt$$

For $R(t)$ flowing with the velocity, the energy transferred to $R(t)$ by the body and surface forces during a short time dt is

$$\left(\int_{R(t)} \vec{f}^B \cdot \vec{v} dV + \int_{dR(t)} \vec{v} \cdot (\sigma \vec{n}) dS \right) dt$$

↙ surface force density

Thus, energy conservation reads (e energy density)

$$\frac{d}{dt} \int_{R(t)} e dV = \int_{R(t)} \vec{f}^B \cdot \vec{v} dV + \int_{dR(t)} \vec{v} \cdot (\sigma \vec{n}) dS$$

$$\Leftrightarrow \int_{R(t)} \left(\frac{de}{dt} + \vec{v} \cdot (e \vec{v}) \right) dV = \int_{R(t)} \left(\vec{f}^B \cdot \vec{v} + \vec{v} \cdot (\sigma \vec{v}) \right) dV,$$

giving

$$\boxed{\frac{de}{dt} + \vec{v} \cdot (e \vec{v}) = \vec{f}^B \cdot \vec{v} + \vec{v} \cdot (\sigma \vec{v})} \quad (10)$$

We can also express (10) in terms of the specific thermal energy density ε : We have

$$e = \frac{1}{2} \rho |\vec{v}|^2 + \rho \varepsilon,$$

Insert this into (10) and express (10) through ε .