

# Chapter 2

## Analysis of Stress

### 2.1 THE CONTINUUM CONCEPT

The molecular nature of the structure of matter is well established. In numerous investigations of material behavior, however, the individual molecule is of no concern and only the behavior of the material as a whole is deemed important. For these cases the observed macroscopic behavior is usually explained by disregarding molecular considerations and, instead, by assuming the material to be continuously distributed throughout its volume and to completely fill the space it occupies. This *continuum concept* of matter is the fundamental postulate of Continuum Mechanics. Within the limitations for which the continuum assumption is valid, this concept provides a framework for studying the behavior of solids, liquids and gases alike.

Adoption of the continuum viewpoint as the basis for the mathematical description of material behavior means that field quantities such as stress and displacement are expressed as piecewise continuous functions of the space coordinates and time.

### 2.2 HOMOGENEITY. ISOTROPY. MASS-DENSITY

A *homogeneous* material is one having identical properties at all points. With respect to some property, a material is *isotropic* if that property is the same in all directions at a point. A material is called *anisotropic* with respect to those properties which are directional at a point.

The concept of *density* is developed from the *mass-volume ratio* in the neighborhood of a point in the continuum. In Fig. 2-1 the mass in the small element of volume  $\Delta V$  is denoted by  $\Delta M$ . The *average density* of the material within  $\Delta V$  is therefore

$$\rho_{(av)} = \frac{\Delta M}{\Delta V} \quad (2.1)$$

The *density* at some interior point  $P$  of the volume element  $\Delta V$  is given mathematically in accordance with the continuum concept by the limit,

$$\rho = \lim_{\Delta V \rightarrow 0} \frac{\Delta M}{\Delta V} = \frac{dM}{dV} \quad (2.2)$$

Mass-density  $\rho$  is a scalar quantity.

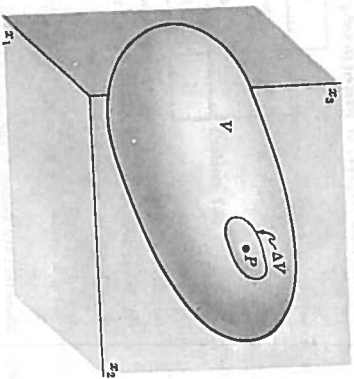


Fig. 2-1

### 2.3 BODY FORCES. SURFACE FORCES

Forces are vector quantities which are best described by intuitive concepts such as push or pull. Those forces which act on all elements of volume of a continuum are known as *body forces*. Examples are gravity and inertia forces. These forces are represented by the symbol  $b_i$  (force per unit mass), or as  $p_i$  (force per unit volume). They are related through the density by the equation

$$\rho b_i = p_i \quad \text{or} \quad \rho b = p \quad (2.3)$$

Those forces which act on a surface element, whether it is a portion of the bounding surface of the continuum or perhaps an arbitrary internal surface, are known as *surface forces*. These are designated by  $f_i$  (force per unit area). Contact forces between bodies are a type of surface forces.

### 2.4 CAUCHY'S STRESS PRINCIPLE. THE STRESS VECTOR

A material continuum occupying the region  $R$  of space, and subjected to surface forces  $f_i$  and body forces  $b_i$ , is shown in Fig. 2-2. As a result of forces being transmitted from one portion of the continuum to another, the material within an arbitrary volume  $V$  enclosed by the surface  $S$  interacts with the material outside of this volume. Taking  $n_i$  as the outward unit normal at point  $P$  of a small element of surface  $\Delta S$  of  $S$ , let  $\Delta f_i$  be the resultant force exerted across  $\Delta S$  upon the material within  $V$  by the material outside of  $V$ . Clearly the force element  $\Delta f_i$  will depend upon the choice of  $\Delta S$  and upon  $n_i$ . It should also be noted that the distribution of force on  $\Delta S$  is not necessarily uniform. Indeed the force distribution is, in general, equipollent to a force and a moment at  $P$ , as shown in Fig. 2-2 by the vectors  $\Delta f_i$  and  $\Delta M_i$ .

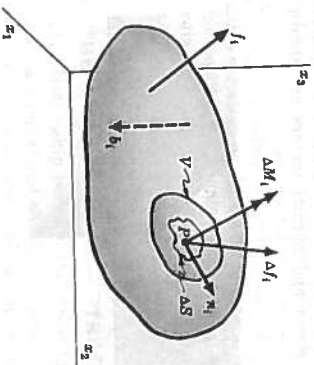


Fig. 2-2

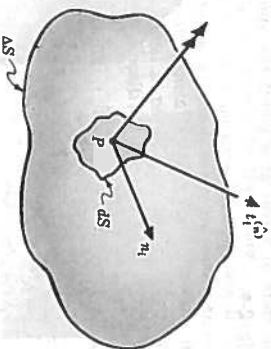


Fig. 2-3

The average force per unit area on  $\Delta S$  is given by  $\Delta f_i / \Delta S$ . The *Cauchy stress principle* asserts that this ratio  $\Delta f_i / \Delta S$  tends to a definite limit  $df_i / dS$  as  $\Delta S$  approaches zero at the point  $P$ , while at the same time the moment of  $\Delta f_i$  about the point  $P$  vanishes in the limiting process. The resulting vector  $df_i / dS$  (force per unit area) is called the *stress vector*  $t_i^{(n)}$ , and is shown in Fig. 2-3. If the moment at  $P$  were not to vanish in the limiting process, a *couple-stress vector*, shown by the double-headed arrow in Fig. 2-3, would also be defined at the point. One branch of the theory of elasticity considers such couple stresses but they are not considered in this text.

Mathematically the stress vector is defined by

$$t_i^{(n)} = \lim_{\Delta S \rightarrow 0} \frac{\Delta f_i}{\Delta S} = \frac{df_i}{dS} \quad \text{or} \quad t_i^{(n)} = \lim_{\Delta S \rightarrow 0} \frac{\Delta f}{\Delta S} = \frac{df}{dS} \quad (2.4)$$

The notation  $t_i^{(n)}$  (or  $t_i^{(n)}$ ) is used to emphasize the fact that the stress vector at a given point  $P$  in the continuum depends explicitly upon the particular surface element  $\Delta S$  chosen there, as represented by the unit normal  $n_i$  (or  $\hat{n}$ ). For some differently oriented surface element, having a different unit normal, the associated stress vector at  $P$  will also be different. The stress vector arising from the action across  $\Delta S$  at  $P$  of the material within  $V$  upon the material outside is the vector  $-t_i^{(n)}$ . Thus by Newton's law of action and reaction,

$$-t_i^{(n)} = t_i^{(-n)} \quad \text{or} \quad -t_i^{(n)} = t_i^{(-n)} \quad (2.5)$$

The stress vector is very often referred to as the traction vector.

### 2.5 STATE OF STRESS AT A POINT. STRESS TENSOR

At an arbitrary point  $P$  in a continuum, Cauchy's stress principle associates a stress vector  $t_i^{(n)}$  with each unit normal vector  $n_i$ , representing the orientation of an infinitesimal surface element having  $P$  as an interior point. This is illustrated in Fig. 2-3. The totality of all possible pairs of such vectors  $t_i^{(n)}$  and  $n_i$  at  $P$  defines the state of stress at that point. Fortunately it is not necessary to specify every pair of stress and normal vectors to completely describe the state of stress at a given point. This may be accomplished by giving the stress vector on each of three mutually perpendicular planes at  $P$ . Coordinate transformation equations then serve to relate the stress vector on any other plane at the point to the given three.

Adopting planes perpendicular to the coordinate axes for the purpose of specifying the state of stress at a point, the appropriate stress and normal vectors are shown in Fig. 2-4.

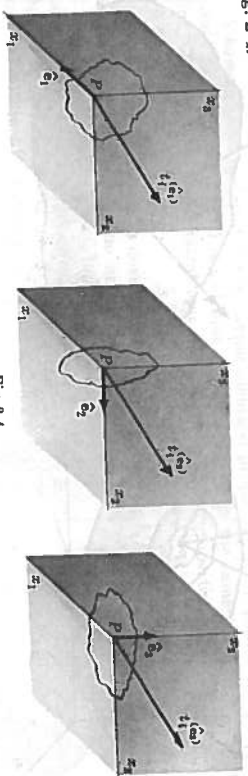


Fig. 2-4

For convenience, the three separate diagrams in Fig. 2-4 are often combined into a single schematic representation as shown in Fig. 2-5 below.

Each of the three coordinate-plane stress vectors may be written according to (1.69) in terms of its Cartesian components as

$$\begin{aligned} t_i^{(e_1)} &= t_1^{(e_1)} e_1 + t_2^{(e_1)} e_2 + t_3^{(e_1)} e_3 = t_j^{(e_1)} e_j \\ t_i^{(e_2)} &= t_1^{(e_2)} e_1 + t_2^{(e_2)} e_2 + t_3^{(e_2)} e_3 = t_j^{(e_2)} e_j \\ t_i^{(e_3)} &= t_1^{(e_3)} e_1 + t_2^{(e_3)} e_2 + t_3^{(e_3)} e_3 = t_j^{(e_3)} e_j \end{aligned} \quad (2.6)$$

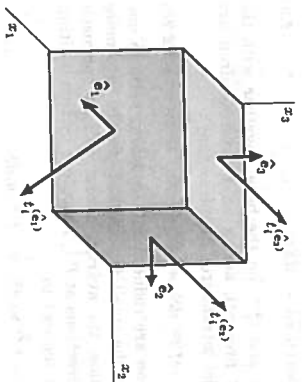


Fig. 2-5

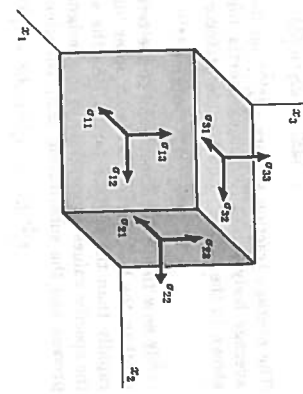


Fig. 2-6

The nine stress vector components,

$$t_j^{(e_i)} = \sigma_{ij} \quad (2.7)$$

are the components of a second-order Cartesian tensor known as the stress tensor. The equivalent stress dyadic is designated by  $\Sigma$ , so that explicit component and matrix representations of the stress tensor, respectively, take the forms

$$\Sigma = \begin{pmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{pmatrix} \quad \text{or} \quad [\sigma_{ij}] = \begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} \quad (2.8)$$

Pictorially, the stress tensor components may be displayed with reference to the coordinate planes as shown in Fig. 2-6. The components perpendicular to the planes ( $\sigma_{11}, \sigma_{22}, \sigma_{33}$ ) are called normal stresses. Those acting in (tangent to) the planes ( $\sigma_{12}, \sigma_{21}, \sigma_{13}, \sigma_{31}, \sigma_{23}, \sigma_{32}$ ) are called shear stresses. A stress component is positive when it acts in the positive direction of the coordinate axes, and on a plane whose outer normal points in one of the positive coordinate directions. The component  $\sigma_{ij}$  acts in the direction of the  $i$ th coordinate axis and on the plane whose outward normal is parallel to the  $j$ th coordinate axis. The stress components shown in Fig. 2-6 are all positive.

### 2.6 THE STRESS TENSOR — STRESS VECTOR RELATIONSHIP

The relationship between the stress tensor  $\sigma_{ij}$  at a point  $P$  and the stress vector  $t_i^{(n)}$  on a plane of arbitrary orientation at that point may be established through the force equilibrium or momentum balance of a small tetrahedron of the continuum, having its vertex at  $P$ . The base of the tetrahedron is taken perpendicular to the coordinate axes as shown by Fig. 2-7. Designating the area of the base  $ABC$  as  $dS$ , the areas of the faces are the projected areas,  $dS_1 = dS n_1$  for face  $CPC$ ,  $dS_2 = dS n_2$  for face  $BPA$  or

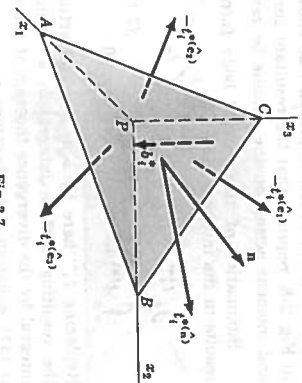


Fig. 2-7

$$dS_1 = dS(\hat{n} \cdot \hat{e}_1) = dS \cos(\hat{n}, \hat{e}_1) = dS n_1 \tag{2.9}$$

The average traction vectors  $-t_i^{(e)}$  on the faces and  $t_i^{(e)}$  on the base, together with the average body forces (including inertia forces, if present), acting on the tetrahedron are shown in the figure. Equilibrium of forces on the tetrahedron requires that

$$t_i^{(e)} dS - t_i^{(e)} dS_1 - t_i^{(e)} dS_2 - t_i^{(e)} dS_3 + \rho b_i^* dV = 0 \tag{2.10}$$

If now the linear dimensions of the tetrahedron are reduced in a constant ratio to one another, the body forces, being an order higher in the small dimensions, tend to zero more rapidly than the surface forces. At the same time, the average stress vectors approach the specific values appropriate to the designated directions at  $P$ . Therefore by this limiting process and the substitution (2.9), equation (2.10) reduces to

$$t_i^{(e)} dS = t_i^{(e)} n_1 dS + t_i^{(e)} n_2 dS + t_i^{(e)} n_3 dS = t_i^{(e)} n_j dS \tag{2.11}$$

Cancelling the common factor  $dS$  and using the identity  $t_i^{(e)} = \sigma_{ji}$ , (2.11) becomes

$$t_i^{(e)} = \sigma_{ji} n_j \quad \text{or} \quad t_i^{(e)} = \hat{n} \cdot \Sigma \tag{2.12}$$

Equation (2.12) is also often expressed in the matrix form

$$[t_i^{(e)}] = [n_{1j}] [\sigma_{ij}] \tag{2.13}$$

which is written explicitly

$$\begin{bmatrix} t_1^{(e)} \\ t_2^{(e)} \\ t_3^{(e)} \end{bmatrix} = \begin{bmatrix} n_1 & n_2 & n_3 \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} \tag{2.14}$$

The matrix form (2.14) is equivalent to the component equations

$$\begin{aligned} t_1^{(e)} &= n_1 \sigma_{11} + n_2 \sigma_{21} + n_3 \sigma_{31} \\ t_2^{(e)} &= n_1 \sigma_{12} + n_2 \sigma_{22} + n_3 \sigma_{32} \\ t_3^{(e)} &= n_1 \sigma_{13} + n_2 \sigma_{23} + n_3 \sigma_{33} \end{aligned} \tag{2.15}$$

### 2.7 FORCE AND MOMENT EQUILIBRIUM. STRESS TENSOR SYMMETRY

Equilibrium of an arbitrary volume  $V$  of a continuum, subjected to a system of surface forces  $t_i^{(e)}$  and body forces  $b_i$  (including inertia forces, if present) as shown in Fig. 2-8, requires that the resultant force and moment acting on the volume be zero.

Summation of surface and body forces results in the integral relation,

$$\begin{aligned} \int_S t_i^{(e)} dS + \int_V \rho b_i dV &= 0 \\ \text{or} \quad \int_S t_i^{(e)} dS + \int_V \rho b_i dV &= 0 \end{aligned} \tag{2.16}$$

Replacing  $t_i^{(e)}$  here by  $\sigma_{ji} n_j$ , and converting the resulting surface integral to a volume integral by the divergence theorem of Gauss (1.157), equation (2.16) becomes

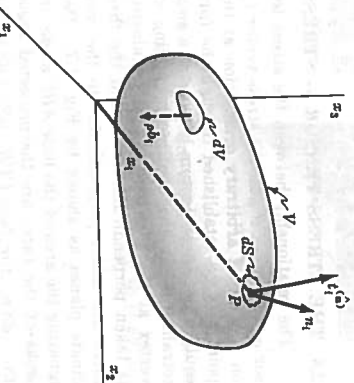


Fig. 2-8

$$\int_V (\sigma_{ji,j} + \rho b_i) dV = 0 \quad \text{or} \quad \int_V (\nabla \cdot \Sigma + \rho b) dV = 0 \tag{2.17}$$

Since the volume  $V$  is arbitrary, the integrand in (2.17) must vanish, so that

$$\sigma_{ji,j} + \rho b_i = 0 \quad \text{or} \quad \nabla \cdot \Sigma + \rho b = 0 \tag{2.18}$$

which are called the *equilibrium equations*.

In the absence of distributed moments or couple-stresses, the equilibrium of moments about the origin requires that

$$\begin{aligned} \int_S \epsilon_{ijk} x_j t_k^{(e)} dS + \int_V \epsilon_{ijk} x_j \rho b_k dV &= 0 \\ \text{or} \quad \int_S \mathbf{x} \times t^{(e)} dS + \int_V \mathbf{x} \times \rho b dV &= 0 \end{aligned} \tag{2.19}$$

in which  $x_i$  is the position vector of the elements of surface and volume. Again, making the substitution  $t_i^{(e)} = \sigma_{ji} n_j$ , applying the theorem of Gauss and using the result expressed in (2.18), the integrals of (2.19) are combined and reduced to

$$\int_V \epsilon_{ijk} \sigma_{jk} dV = 0 \quad \text{or} \quad \int_V \Sigma_{ij} dV = 0 \tag{2.20}$$

For the arbitrary volume  $V$ , (2.20) requires

$$\epsilon_{ijk} \sigma_{jk} = 0 \quad \text{or} \quad \Sigma_{ij} = 0 \tag{2.21}$$

Equation (2.21) represents the equations  $\sigma_{12} = \sigma_{21}$ ,  $\sigma_{23} = \sigma_{32}$ ,  $\sigma_{31} = \sigma_{13}$ , or in all

$$\sigma_{ij} = \sigma_{ji} \tag{2.22}$$

which shows that the *stress tensor is symmetric*. In view of (2.22), the equilibrium equations (2.18) are often written

$$\sigma_{ij,j} + \rho b_i = 0 \tag{2.23}$$

which appear in expanded form as

$$\begin{aligned} \frac{\partial \sigma_{11}}{\partial x_1} + \frac{\partial \sigma_{12}}{\partial x_2} + \frac{\partial \sigma_{13}}{\partial x_3} + \rho b_1 &= 0 \\ \frac{\partial \sigma_{21}}{\partial x_1} + \frac{\partial \sigma_{22}}{\partial x_2} + \frac{\partial \sigma_{23}}{\partial x_3} + \rho b_2 &= 0 \\ \frac{\partial \sigma_{31}}{\partial x_1} + \frac{\partial \sigma_{32}}{\partial x_2} + \frac{\partial \sigma_{33}}{\partial x_3} + \rho b_3 &= 0 \end{aligned} \tag{2.24}$$

### 2.8 STRESS TRANSFORMATION LAWS

At the point  $P$  let the rectangular Cartesian coordinate systems  $Px_1x_2x_3$  and  $Px'_1x'_2x'_3$  of Fig. 2-9 be related to one another by the table of direction cosines

	$x_1$	$x_2$	$x_3$
$x'_1$	$a_{11}$	$a_{12}$	$a_{13}$
$x'_2$	$a_{21}$	$a_{22}$	$a_{23}$
$x'_3$	$a_{31}$	$a_{32}$	$a_{33}$

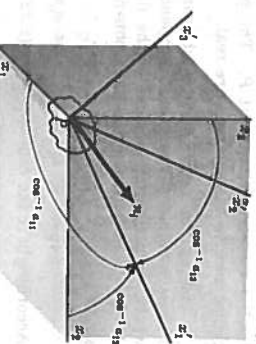


Fig. 2-9

or by the equivalent alternatives, the transformation matrix  $[a_{ij}]$  or the transformation dyadic

$$A = a_{ij}e_i e_j \quad (2.25)$$

According to the transformation law for Cartesian tensors of order one (1.32), the components of the stress vector  $t_i^{(\hat{n})}$  referred to the unprimed axes are related to the primed axes components  $t_i^{(\hat{n})}$  by the equation

$$t_i^{(\hat{n})} = a_{ij} t_j^{(\hat{n})} \quad \text{or} \quad t_i^{(\hat{n})} = A \cdot t^{(\hat{n})} \quad (2.26)$$

Likewise, by the transformation law (1.102) for second-order Cartesian tensors, the stress tensor components in the two systems are related by

$$\sigma'_{ij} = a_{ip} a_{jq} \sigma_{pq} \quad \text{or} \quad \Sigma' = A \cdot \Sigma \cdot A \quad (2.27)$$

In matrix form, the stress vector transformation is written

$$[t_i^{(\hat{n})}] = [a_{ij}] [t_j^{(\hat{n})}] \quad (2.28)$$

and the stress tensor transformation as

$$[\sigma'_{ij}] = [a_{ip}] [\sigma_{pq}] [a_{jq}] \quad (2.29)$$

Explicitly, the matrix multiplications in (2.28) and (2.29) are given respectively by

$$\begin{bmatrix} t_1^{(\hat{n})} \\ t_2^{(\hat{n})} \\ t_3^{(\hat{n})} \end{bmatrix} = \begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{21} & a_{22} & a_{23} \\ a_{31} & a_{32} & a_{33} \end{bmatrix} \begin{bmatrix} t_1^{(\hat{n})} \\ t_2^{(\hat{n})} \\ t_3^{(\hat{n})} \end{bmatrix} \quad (2.30)$$

and

$$\begin{bmatrix} \sigma'_{11} & \sigma'_{12} & \sigma'_{13} \\ \sigma'_{21} & \sigma'_{22} & \sigma'_{23} \\ \sigma'_{31} & \sigma'_{32} & \sigma'_{33} \end{bmatrix} = \begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{21} & a_{22} & a_{23} \\ a_{31} & a_{32} & a_{33} \end{bmatrix} \begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} \begin{bmatrix} a_{11} & a_{21} & a_{31} \\ a_{12} & a_{22} & a_{32} \\ a_{13} & a_{23} & a_{33} \end{bmatrix} \quad (2.31)$$

### 2.9 STRESS QUADRIC OF CAUCHY

At the point  $P$  in a continuum, let the stress tensor have the values  $\sigma_{ij}$  when referred to directions parallel to the local Cartesian axes  $P_1, P_2, P_3$  shown in Fig. 2-10. The equation

$$\sigma_{ij} \xi_i \xi_j = \pm k^2 \quad (2.32)$$

represents geometrically similar quadric surfaces having a common center at  $P$ . The plus or minus choice assures the surfaces are real.

The position vector  $r$  of an arbitrary point lying on the quadric surface has components  $\xi_i = r n_i$ , where  $n_i$  is the unit normal in the direction of  $r$ . At the point  $P$  the normal component  $\sigma_n n_i$  of the stress vector  $t_i^{(\hat{n})}$  has a magnitude

$$\sigma_n = t_i^{(\hat{n})} n_i = t_i^{(\hat{n})} \cdot n = \sigma_{ij} n_j n_i \quad (2.33)$$

Accordingly if the constant  $k^2$  of (2.32) is set equal to  $\sigma_n n^2$ , the resulting quadric

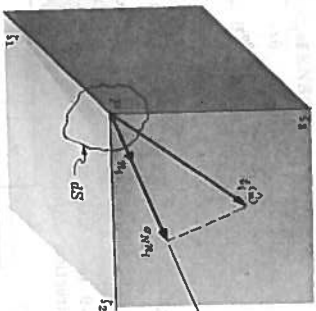
$$\sigma_{ij} \xi_i \xi_j = \pm \sigma_n n^2 \quad (2.34)$$


Fig. 2-10

is called the *stress quadric of Cauchy*. From this definition it follows that the magnitude  $\sigma_n$  of the normal stress component on the surface element  $dS$  perpendicular to the position vector  $r$  of a point on Cauchy's stress quadric, is inversely proportional to  $r^2$ , i.e.  $\sigma_n = \pm k^2/r^2$ . Furthermore it may be shown that the stress vector  $t_i^{(\hat{n})}$  acting on  $dS$  at  $P$  is parallel to the normal of the tangent plane of the Cauchy quadric at the point identified by  $r$ .

### 2.10 PRINCIPAL STRESSES. STRESS INVARIANTS. STRESS ELLIPSOID

At the point  $P$  for which the stress tensor components are  $\sigma_{ij}$ , the equation (2.12),  $t_i^{(\hat{n})} = \sigma_{ij} n_j$ , associates with each direction  $n_i$  a stress vector  $t_i^{(\hat{n})}$ . Those directions for which  $t_i^{(\hat{n})}$  and  $n_i$  are collinear as shown in Fig. 2-11 are called *principal stress directions*. For a principal stress direction,

$$t_i^{(\hat{n})} = \sigma n_i \quad \text{or} \quad t_i^{(\hat{n})} = \sigma \hat{n} \quad (2.35)$$

in which  $\sigma$ , the magnitude of the stress vector, is called a *principal stress value*. Substituting (2.35) into (2.12) and making use of the identities  $n_i = \delta_{ij} n_j$  and  $\sigma_{ij} = \sigma_{ji}$ , results in the equations

$$(\sigma_{ij} - \delta_{ij} \sigma) n_j = 0 \quad \text{or} \quad (\Sigma - I \sigma) \cdot \hat{n} = 0 \quad (2.36)$$

In the three equations (2.36), there are four unknowns, namely, the three direction cosines  $n_i$  and the principal stress value  $\sigma$ .

For solutions of (2.36) other than the trivial one  $n_i = 0$ , the determinant of coefficients,  $|\sigma_{ij} - \delta_{ij} \sigma|$ , must vanish. Explicitly,

$$\begin{vmatrix} \sigma_{11} - \sigma & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} - \sigma & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} - \sigma \end{vmatrix} = 0 \quad (2.37)$$

which upon expansion yields the cubic polynomial in  $\sigma$ ,

$$\sigma^3 - I_1 \sigma^2 + II_1 \sigma - III_1 = 0 \quad (2.38)$$

$$I_1 = \sigma_{ii} = \text{tr } \Sigma \quad (2.39)$$

$$II_1 = \frac{1}{2}(\sigma_{ij} \sigma_{ji} - \sigma_{ij} \sigma_{ij}) \quad (2.40)$$

$$III_1 = |\sigma_{ij}| = \det \Sigma \quad (2.41)$$

are known respectively as the *first, second and third stress invariants*.

The three roots of (2.38),  $\sigma^{(1)}, \sigma^{(2)}, \sigma^{(3)}$ , are the three principal stress values. Associated with each principal stress  $\sigma^{(k)}$ , there is a principal stress direction for which the direction cosines  $n_i^{(k)}$  are solutions of the equations

$$(\sigma_{ij} - \sigma^{(k)} \delta_{ij}) n_j^{(k)} = 0 \quad \text{or} \quad (\Sigma - \sigma^{(k)} I) \cdot \hat{n}^{(k)} = 0 \quad (k = 1, 2, 3) \quad (2.42)$$

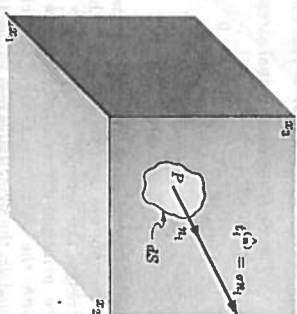


Fig. 2-11

In (2.42) letter subscripts or superscripts enclosed by parentheses are merely labels and as such do not participate in any summation process. The expanded form of (2.42) for the second principal direction, for example, is therefore

$$\begin{aligned} (\sigma_{11} - \sigma_{(2)})n_1^{(2)} + \sigma_{12}n_2^{(2)} + \sigma_{13}n_3^{(2)} &= 0 \\ \sigma_{21}n_1^{(2)} + (\sigma_{22} - \sigma_{(2)})n_2^{(2)} + \sigma_{23}n_3^{(2)} &= 0 \\ \sigma_{31}n_1^{(2)} + \sigma_{32}n_2^{(2)} + (\sigma_{33} - \sigma_{(2)})n_3^{(2)} &= 0 \end{aligned} \quad (2.43)$$

Because the stress tensor is real and symmetric, the principal stress values are also real. When referred to principal stress directions, the stress matrix  $[\sigma_{ij}]$  is diagonal,

$$[\sigma_{ij}] = \begin{bmatrix} \sigma_{(1)} & 0 & 0 \\ 0 & \sigma_{(2)} & 0 \\ 0 & 0 & \sigma_{(3)} \end{bmatrix} \quad \text{or} \quad [\sigma_{ij}] = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_{II} & 0 \\ 0 & 0 & \sigma_{III} \end{bmatrix} \quad (2.44)$$

in the second form of which Roman numeral subscripts are used to show that the principal stresses are ordered, i.e.  $\sigma_1 > \sigma_{II} > \sigma_{III}$ . Since the principal stress directions are coincident with the principal axes of Cauchy's stress quadric, the principal stress values include both the maximum and minimum normal stress components at a point.

In a principal stress space, i.e. a space whose axes are in the principal stress directions and whose coordinate unit of measure is stress ( $t_1^{(\hat{n})}, t_2^{(\hat{n})}, t_3^{(\hat{n})}$ ) as shown in Fig. 2-12, the arbitrary stress vector  $t_i^{(\hat{n})}$  has components

$$t_i^{(\hat{n})} = \sigma_{(1)}n_i, \quad t_2^{(\hat{n})} = \sigma_{(2)}n_i, \quad t_3^{(\hat{n})} = \sigma_{(3)}n_i \quad (2.45)$$

according to (2.12). But inasmuch as  $(n_1)^2 + (n_2)^2 + (n_3)^2 = 1$  for the unit vector  $n_i$ , (2.45) requires the stress vector  $t_i^{(\hat{n})}$  to satisfy the equation

$$\frac{(t_1^{(\hat{n})})^2}{(\sigma_{(1)})^2} + \frac{(t_2^{(\hat{n})})^2}{(\sigma_{(2)})^2} + \frac{(t_3^{(\hat{n})})^2}{(\sigma_{(3)})^2} = 1 \quad (2.46)$$

in stress space. This equation is an ellipsoid known as the Lamé stress ellipsoid.

### 2.11 MAXIMUM AND MINIMUM SHEAR STRESS VALUES

If the stress vector  $t_i^{(\hat{n})}$  is resolved into orthogonal components normal and tangential to the surface element  $dS$  upon which it acts, the magnitude of the normal component may be determined from (2.33) and the magnitude of the tangential or shearing component is given by

$$\sigma_s^2 = t_i^{(\hat{n})}t_i^{(\hat{n})} - \sigma_n^2 \quad (2.47)$$

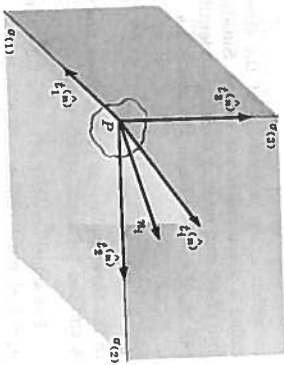


Fig. 2-12

This resolution is shown in Fig. 2-13 where the axes are chosen in the principal stress directions and it is assumed the principal stresses are ordered according to  $\sigma_1 > \sigma_{II} > \sigma_{III}$ . Hence from (2.12), the components of  $t_i^{(\hat{n})}$  are

$$\begin{aligned} t_1^{(\hat{n})} &= \sigma_1 n_1 \\ t_2^{(\hat{n})} &= \sigma_{II} n_2 \\ t_3^{(\hat{n})} &= \sigma_{III} n_3 \end{aligned} \quad (2.48)$$

and from (2.33), the normal component magnitude is

$$\sigma_n = \sigma_1 n_1^2 + \sigma_{II} n_2^2 + \sigma_{III} n_3^2 \quad (2.49)$$

Substituting (2.48) and (2.49) into (2.47), the squared magnitude of the shear stress as a function of the direction cosines  $n_i$  is given by

$$\sigma_s^2 = \sigma_1^2 n_1^2 + \sigma_{II}^2 n_2^2 + \sigma_{III}^2 n_3^2 - (\sigma_1 n_1^2 + \sigma_{II} n_2^2 + \sigma_{III} n_3^2)^2 \quad (2.50)$$

The maximum and minimum values of  $\sigma_s$  may be obtained from (2.50) by the method of Lagrangian multipliers. The procedure is to construct the function

$$F = \sigma_s^2 - \lambda n_i n_i \quad (2.51)$$

in which the scalar  $\lambda$  is called a Lagrangian multiplier. Equation (2.51) is clearly a function of the direction cosines  $n_i$ , so that the conditions for stationary (maximum or minimum) values of  $F$  are given by  $\partial F / \partial n_i = 0$ . Setting these partials equal to zero yields the equations

$$n_1 \{ \sigma_1^2 - 2\sigma_1(\sigma_1 n_1^2 + \sigma_{II} n_2^2 + \sigma_{III} n_3^2) + \lambda \} = 0 \quad (2.52a)$$

$$n_2 \{ \sigma_{II}^2 - 2\sigma_{II}(\sigma_1 n_1^2 + \sigma_{II} n_2^2 + \sigma_{III} n_3^2) + \lambda \} = 0 \quad (2.52b)$$

$$n_3 \{ \sigma_{III}^2 - 2\sigma_{III}(\sigma_1 n_1^2 + \sigma_{II} n_2^2 + \sigma_{III} n_3^2) + \lambda \} = 0 \quad (2.52c)$$

which, together with the condition  $n_i n_i = 1$ , may be solved for  $\lambda$  and the direction cosines  $n_1, n_2, n_3$ , conjugate to the extremum values of shear stress.

One set of solutions to (2.52), and the associated shear stresses from (2.50), are

$$n_1 = \pm 1, \quad n_2 = 0, \quad n_3 = 0; \quad \text{for which } \sigma_s = 0 \quad (2.53a)$$

$$n_1 = 0, \quad n_2 = \pm 1, \quad n_3 = 0; \quad \text{for which } \sigma_s = 0 \quad (2.53b)$$

$$n_1 = 0, \quad n_2 = 0, \quad n_3 = \pm 1; \quad \text{for which } \sigma_s = 0 \quad (2.53c)$$

The shear stress values in (2.53) are obviously minimum values. Furthermore, since (2.53) indicates that shear components vanish on principal planes, the directions given by (2.53) are recognized as principal stress directions.

A second set of solutions to (2.52) may be verified to be given by

$$n_1 = 0, \quad n_2 = \pm 1/\sqrt{2}, \quad n_3 = \pm 1/\sqrt{2}; \quad \text{for which } \sigma_s = (\sigma_{II} - \sigma_{III})/2 \quad (2.54a)$$

$$n_1 = \pm 1/\sqrt{2}, \quad n_2 = 0, \quad n_3 = \pm 1/\sqrt{2}; \quad \text{for which } \sigma_s = (\sigma_{III} - \sigma_1)/2 \quad (2.54b)$$

$$n_1 = \pm 1/\sqrt{2}, \quad n_2 = \pm 1/\sqrt{2}, \quad n_3 = 0; \quad \text{for which } \sigma_s = (\sigma_1 - \sigma_{II})/2 \quad (2.54c)$$

Equation (2.54b) gives the maximum shear stress value, which is equal to half the difference of the largest and smallest principal stresses. Also from (2.54b), the maximum shear stress component acts in the plane which bisects the right angle between the directions of the maximum and minimum principal stresses.

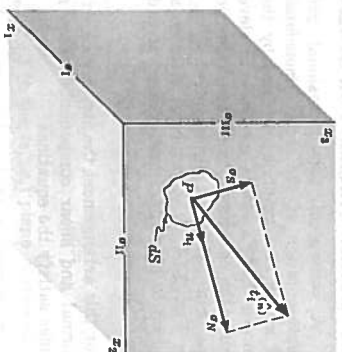


Fig. 2-13

2.12 MOHR'S CIRCLES FOR STRESS

A convenient two-dimensional graphical representation of the three-dimensional state of stress at a point is provided by the well-known Mohr's stress circles. In developing these, the coordinate axes are again chosen in the principal stress directions at  $P$  as shown by Fig. 2-14. The principal stresses are assumed to be distinct and ordered according to

$$\sigma_1 > \sigma_{II} > \sigma_{III} \quad (2.55)$$

For this arrangement the stress vector  $t_i^{(n)}$  has normal and shear components whose magnitudes satisfy the equations

$$\sigma_N = \sigma_1 n_1^2 + \sigma_{II} n_2^2 + \sigma_{III} n_3^2 \quad (2.56)$$

$$\sigma_N^2 + \sigma_s^2 = \sigma_1^2 n_1^2 + \sigma_{II}^2 n_2^2 + \sigma_{III}^2 n_3^2 \quad (2.57)$$

Combining these two expressions with the identity  $n_1 n_2 = 1$  and solving for the direction cosines  $n_i$ , results in the equations

$$(n_1)^2 = \frac{(\sigma_N - \sigma_{III})(\sigma_N - \sigma_{II}) + (\sigma_s)^2}{(\sigma_1 - \sigma_{III})(\sigma_1 - \sigma_{II})} \quad (2.58a)$$

$$(n_2)^2 = \frac{(\sigma_N - \sigma_{III})(\sigma_N - \sigma_1) + (\sigma_s)^2}{(\sigma_{II} - \sigma_{III})(\sigma_{II} - \sigma_1)} \quad (2.58b)$$

$$(n_3)^2 = \frac{(\sigma_N - \sigma_1)(\sigma_N - \sigma_{II}) + (\sigma_s)^2}{(\sigma_{III} - \sigma_1)(\sigma_{III} - \sigma_{II})} \quad (2.58c)$$

These equations serve as the basis for Mohr's stress circles, shown in the "stress plane" of Fig. 2-15, for which the  $\sigma_N$  axis is the abscissa, and the  $\sigma_s$  axis is the ordinate.

In (2.58a), since  $\sigma_1 - \sigma_{III} > 0$  and  $\sigma_1 - \sigma_{II} > 0$  from (2.55), and since  $(n_1)^2$  is non-negative, the numerator of the right-hand side satisfies the relationship

$$(\sigma_N - \sigma_1)(\sigma_N - \sigma_{III}) + (\sigma_s)^2 \geq 0 \quad (2.59)$$

which represents stress points in the  $(\sigma_N, \sigma_s)$  plane that are on or exterior to the circle

$$[\sigma_N - (\sigma_{II} + \sigma_{III})/2]^2 + (\sigma_s)^2 = [(\sigma_{II} - \sigma_{III})/2]^2 \quad (2.60)$$

In Fig. 2-15, this circle is labeled  $C_1$ .

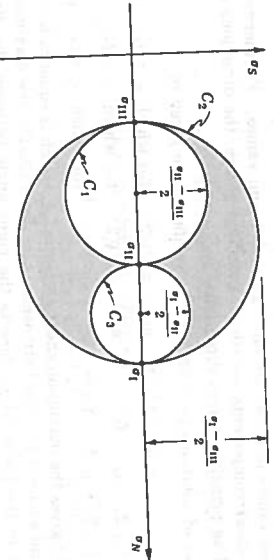


Fig. 2-15

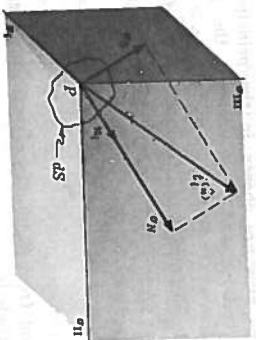


Fig. 2-14

Similarly, for (2.58b), since  $\sigma_{II} - \sigma_{III} > 0$  and  $\sigma_{II} - \sigma_1 < 0$  from (2.55), and since  $(n_2)^2$  is non-negative, the right hand numerator satisfies

$$(\sigma_N - \sigma_{III})(\sigma_N - \sigma_1) + (\sigma_s)^2 \leq 0 \quad (2.61)$$

which represents points on or interior to the circle

$$[\sigma_N - (\sigma_{II} + \sigma_1)/2]^2 + (\sigma_s)^2 = [(\sigma_{II} - \sigma_1)/2]^2 \quad (2.62)$$

labeled  $C_2$  in Fig. 2-15. Finally, for (2.58c), since  $\sigma_{III} - \sigma_1 < 0$  and  $\sigma_{III} - \sigma_{II} < 0$  from (2.55), and since  $(n_3)^2$  is non-negative,

$$(\sigma_N - \sigma_1)(\sigma_N - \sigma_{II}) + (\sigma_s)^2 \geq 0 \quad (2.63)$$

which represents points on or exterior to the circle

$$[\sigma_N - (\sigma_1 + \sigma_{II})/2]^2 + (\sigma_s)^2 = [(\sigma_1 - \sigma_{II})/2]^2 \quad (2.64)$$

labeled  $C_3$  in Fig. 2-15.

Since each "stress point" (pair of values of  $\sigma_N$  and  $\sigma_s$ ) in the  $(\sigma_N, \sigma_s)$  plane represents a particular stress vector  $t_i^{(n)}$ , the state of stress at  $P$  expressed by (2.58) is represented in Fig. 2-15 as the shaded area bounded by the Mohr's stress circles. The diagram confirms a maximum shear stress of  $(\sigma_1 - \sigma_{III})/2$  as was determined analytically in Section 2.11. Frequently, because the sign of the shear stress is not of critical importance, only the top half of this symmetrical diagram is drawn.

The relationship between Mohr's stress diagram and the physical state of stress may be established through consideration of Fig. 2-16, which shows the first octant of a sphere of the continuum centered at point  $P$ . The normal  $n_i$  at the arbitrary point  $Q$  of the spherical surface  $ABC$  simulates the normal to the surface element  $dS$  at point  $P$ . Because of the symmetry properties of the stress tensor and the fact that principal stress axes are used in Fig. 2-16, the state of stress at  $P$  is completely represented through the totality of locations  $Q$  can occupy on the surface  $ABC$ . In the figure, circle arcs  $KD$ ,  $GE$  and  $FH$  designate locations for  $Q$  along which one direction cosine of  $n_i$  has a constant value. Specifically,

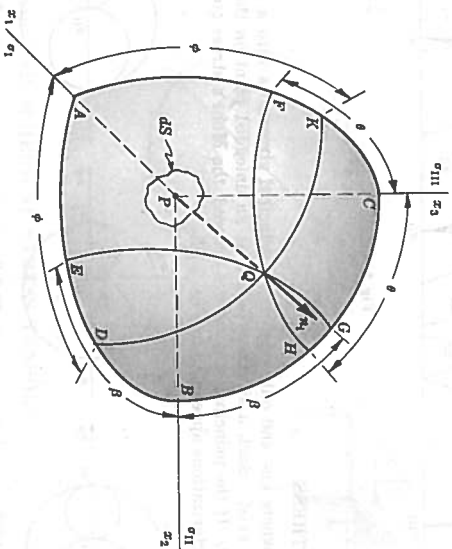


Fig. 2-16

and, on the bounding circle arcs  $BC$ ,  $CA$  and  $AB$ ,  
 $\tau_1 = \cos \phi$  on  $KD$ ,  $\tau_2 = \cos \beta$  on  $GE$ ,  $\tau_3 = \cos \theta$  on  $FH$

$$\tau_1 = \cos \pi/2 = 0 \text{ on } BC, \quad \tau_2 = \cos \pi/2 = 0 \text{ on } CA, \quad \tau_3 = \cos \pi/2 = 0 \text{ on } AB$$

According to the first of these and the equation (2.58a), stress vectors for  $Q$  located on  $BC$  will have components given by stress points on the circle  $C_1$  in Fig. 2-15. Likewise,  $CA$  in Fig. 2-16 corresponds to the circle  $C_2$  and  $AB$  to the circle  $C_3$  in Fig. 2-15.

The stress vector components  $\sigma'_N$  and  $\sigma'_s$  for an arbitrary location of  $Q$  may be determined by the construction shown in Fig. 2-17. Thus point  $e$  may be located on  $C_3$  by drawing the radial line from the center of  $C_3$  at the angle  $2\theta$ . Note that angles in the physical space of Fig. 2-16 are doubled in the stress space of Fig. 2-17 (arc  $AB$  subtends  $90^\circ$  in Fig. 2-16 whereas the conjugate stress points  $\sigma_1$  and  $\sigma_{II}$  are  $180^\circ$  apart on  $C_3$ ). In the same way, points  $g$ ,  $h$  and  $f$  are located in Fig. 2-17 and the appropriate pairs joined by circle arcs having their centers on the  $\sigma'_N$  axis. The intersection of circle arcs  $ge$  and  $hf$  represents the components  $\sigma'_N$  and  $\sigma'_s$  of the stress vector  $t'_i(\hat{n})$  on the plane having the normal direction  $\hat{n}$  at  $Q$  in Fig. 2-16.

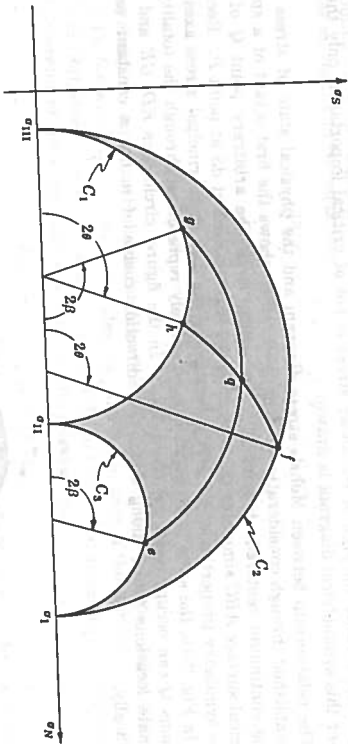


FIG. 2-17

2.13 PLANE STRESS

In the case where one and only one of the principal stresses is zero a state of *plane stress* is said to exist. Such a situation occurs at an unloaded point on the free surface bounding a body. If the principal stresses are ordered, the Mohr's stress circles will have one of the characteristics appearing in Fig. 2-18.

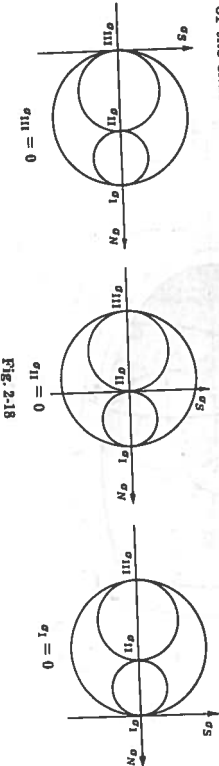


FIG. 2-18

If the principal stresses are not ordered and the direction of the zero principal stress is taken as the  $x_3$  direction, the state of stress is termed *plane stress* parallel to the  $x_1x_2$  plane. For arbitrary choice of orientation of the orthogonal axes  $x_1$  and  $x_2$  in this case, the stress matrix has the form

$$[\sigma_{ij}] = \begin{bmatrix} \sigma_{11} & \sigma_{12} & 0 \\ \sigma_{12} & \sigma_{22} & 0 \\ 0 & 0 & 0 \end{bmatrix} \tag{2.65}$$

The stress quadratic for this plane stress is a cylinder with its base lying in the  $x_1x_2$  plane and having the equation

$$\sigma_{11}x_1^2 + 2\sigma_{12}x_1x_2 + \sigma_{22}x_2^2 = \pm k^2 \tag{2.66}$$

Frequently in elementary books on Strength of Materials a state of plane stress is represented by a single Mohr's circle. As seen from Fig. 2-18 this representation is necessarily incomplete since all three circles are required to show the complete stress picture. In particular, the maximum shear stress value at a point will not be given if the single circle presented happens to be one of the inner circles of Fig. 2-18. A single circle Mohr's diagram is able, however, to display the stress points for all those planes at the point  $P$  which include the zero principal stress axis. For such planes, if the coordinate axes are chosen in accordance with the stress representation given in (2.65), the single plane stress Mohr's circle has the equation

$$[\sigma'_N - (\sigma'_{11} + \sigma'_{22})/2]^2 + (\sigma'_s)^2 = [(\sigma'_{11} - \sigma'_{22})/2]^2 + (\sigma'_{12})^2 \tag{2.67}$$

The essential features in the construction of this circle are illustrated in Fig. 2-19. The circle is drawn by locating the center  $C$  at  $\sigma'_N = (\sigma'_{11} + \sigma'_{22})/2$  and using the radius  $R = \sqrt{[(\sigma'_{11} - \sigma'_{22})/2]^2 + (\sigma'_{12})^2}$  given in (2.67). Point  $A$  on the circle represents the stress state on the surface element whose normal is  $\hat{n}_1$  (the right-hand face of the rectangular parallelepiped shown in Fig. 2-19). Point  $B$  on the circle represents the stress state on the top surface of the parallelepiped with normal  $\hat{n}_2$ . Principal stress points  $\sigma_1$  and  $\sigma_{II}$  are so labeled, and points  $E$  and  $D$  on the circle are points of maximum shear stress value.

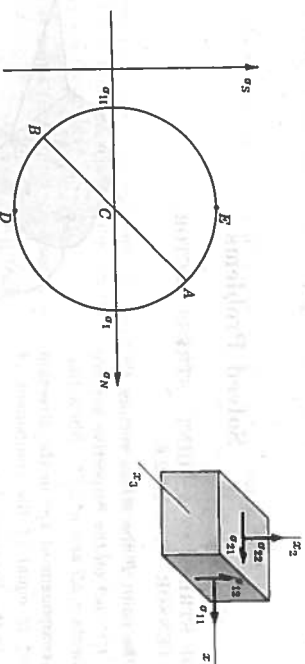


FIG. 2-19

2.14 DEVIATOR AND SPHERICAL STRESS TENSORS

It is very often useful to split the stress tensor  $\sigma_{ij}$  into two component tensors, one of which (the *spherical* or *hydrostatic stress tensor*) has the form

$$\Sigma_M = \sigma_M \mathbf{1} = \begin{pmatrix} \sigma_M & 0 & 0 \\ 0 & \sigma_M & 0 \\ 0 & 0 & \sigma_M \end{pmatrix} \quad (2.68)$$

where  $\sigma_M = -p = \sigma_{kk}/3$  is the mean normal stress, and the second (the deviator stress tensor) has the form

$$\Sigma_D = \begin{pmatrix} \sigma_{11} - \sigma_M & \sigma_{12} & \sigma_{13} \\ \sigma_{21} - \sigma_M & \sigma_{22} - \sigma_M & \sigma_{23} \\ \sigma_{31} - \sigma_M & \sigma_{32} - \sigma_M & \sigma_{33} - \sigma_M \end{pmatrix} \equiv \begin{pmatrix} s_{11} & s_{12} & s_{13} \\ s_{21} & s_{22} & s_{23} \\ s_{31} & s_{32} & s_{33} \end{pmatrix} \quad (2.69)$$

This decomposition is expressed by the equations

$$\sigma_{ij} = s_{ij} + \sigma_M \delta_{ij} \quad \text{or} \quad \Sigma = \Sigma_M + \Sigma_D \quad (2.70)$$

The principal directions of the deviator stress tensor  $s_{ij}$  are the same as those of the stress tensor  $\sigma_{ij}$ . Thus principal deviator stress values are

$$s_{(k)} = \sigma_{(k)} - \sigma_M \quad (2.71)$$

The characteristic equation for the deviator stress tensor, comparable to (2.58) for the stress tensor, is the cubic

$$s^3 + \text{II} s - \text{III} = 0 \quad \text{or} \quad s^3 + (s_{11}s_{22} + s_{11}s_{33} + s_{22}s_{33})s - s_{12}s_{21} - s_{13}s_{31} - s_{23}s_{32} = 0 \quad (2.72)$$

It is easily shown that the first invariant of the deviator stress tensor  $\text{III}$  is identically zero, which accounts for its absence in (2.72).

### Solved Problems

#### STATE OF STRESS AT A POINT. STRESS VECTOR.

##### 2.1. STRESS TENSOR (Sec. 2.1.2.6)

At the point  $P$  the stress vectors  $t_i^{(j)}$  and  $t_j^{(i)}$  act on the respective surface elements  $\Delta S$  and  $\Delta S^*$ . Show that the component of  $t_i^{(j)}$  in the direction of  $n_i^*$  is equal to the component of  $t_j^{(i)}$  in the direction of  $n_j$ .

It is required to show that

$$t_i^{(j)} n_i^* = t_j^{(i)} n_j$$

From (2.12)  $t_i^{(j)} n_i = \sigma_{ij} n_j$ , and by (2.22)  $\sigma_{ji} = \sigma_{ij}$ , so that

$$\sigma_{ij} n_j^* n_i = (\sigma_{ij} n_j)^* = t_j^{(i)} n_j^*$$

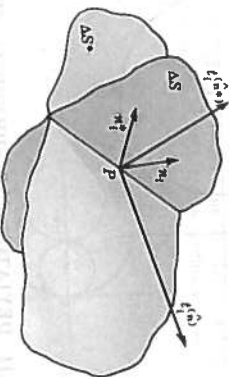


FIG. 2-20

##### 2.2. The stress tensor values at a point $P$ are given by the array

$$\Sigma = \begin{pmatrix} 7 & 0 & -2 \\ 0 & 5 & 0 \\ -2 & 0 & 4 \end{pmatrix}$$

Determine the traction (stress) vector on the plane at  $P$  whose unit normal is  $\hat{n} = (2/3)\hat{e}_1 - (2/3)\hat{e}_2 + (1/3)\hat{e}_3$ .

From (2.12),  $t^{(j)} = \hat{n} \cdot \Sigma$ . The multiplication is best carried out in the matrix form of (2.12):

$$\begin{bmatrix} t_1^{(j)} \\ t_2^{(j)} \\ t_3^{(j)} \end{bmatrix} = \begin{bmatrix} 2/3 & -2/3 & 1/3 \\ 0 & 5 & 0 \\ -2 & 0 & 4 \end{bmatrix} = \begin{bmatrix} 14/3 & -2/3 & -4/3 + 4 \\ 0 & 5 & 0 \\ -2 & 0 & 4 \end{bmatrix}$$

Thus  $t^{(j)} = 4\hat{e}_1 - \frac{10}{3}\hat{e}_2$ .

##### 2.3. For the traction vector of Problem 2.2, determine (a) the component perpendicular to the plane, (b) the magnitude of $t^{(j)}$ , (c) the angle between $t^{(j)}$ and $\hat{n}$ .

(a)  $t^{(j)} \cdot \hat{n} = (4\hat{e}_1 - \frac{10}{3}\hat{e}_2) \cdot (\frac{2}{3}\hat{e}_1 - \frac{2}{3}\hat{e}_2 + \frac{1}{3}\hat{e}_3) = 44/9$

(b)  $|t^{(j)}| = \sqrt{16 + 100/9} = 5.2$

(c) Since  $t^{(j)} \cdot \hat{n} = |t^{(j)}| \cos \theta$ ,  $\cos \theta = (44/9)/(5.2) = 0.94$  and  $\theta = 20^\circ$ .

##### 2.4. The stress vectors acting on the three coordinate planes are given by $t_i^{(1)}$ , $t_i^{(2)}$ and $t_i^{(3)}$ . Show that the sum of the squares of the magnitudes of these vectors is independent of the orientation of the coordinate planes.

Let  $S$  be the sum in question. Then

$$S = t_i^{(1)} t_i^{(1)} + t_i^{(2)} t_i^{(2)} + t_i^{(3)} t_i^{(3)}$$

which from (2.7) becomes  $S = \sigma_{11}^2 + \sigma_{22}^2 + \sigma_{33}^2 + \sigma_{21}^2 + \sigma_{12}^2 + \sigma_{31}^2 + \sigma_{13}^2$ , an invariant.

##### 2.5. The state of stress at a point is given by the stress tensor

$$\sigma_{ij} = \begin{pmatrix} a & a & b \\ a & a & c \\ b & c & c \end{pmatrix}$$

where  $a, b, c$  are constants and  $\sigma$  is some stress value. Determine the constants  $a, b$  and  $c$  so that the stress vector on the octahedral plane ( $\hat{n} = (1/\sqrt{3})\hat{e}_1 + (1/\sqrt{3})\hat{e}_2 + (1/\sqrt{3})\hat{e}_3$ ) vanishes.

In matrix form,  $t_i^{(j)} = \sigma_{ij} n_j$ , must be zero for the given stress tensor and normal vector.

$$\begin{bmatrix} a & a & b \\ a & a & c \\ b & c & c \end{bmatrix} \begin{bmatrix} 1/\sqrt{3} \\ 1/\sqrt{3} \\ 1/\sqrt{3} \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} \quad \text{hence} \quad \begin{matrix} a + b = -1 \\ a + c = -1 \\ b + c = -1 \end{matrix}$$

Solving these equations,  $a = b = c = -1/2$ . Therefore the solution tensor is

$$\sigma_{ij} = \begin{pmatrix} -1/2 & -1/2 & -1/2 \\ -1/2 & -1/2 & -1/2 \\ -1/2 & -1/2 & -1/2 \end{pmatrix}$$